

RTCA, Inc.  
1150 18<sup>th</sup> Street, NW, Suite 910  
Washington, DC 20036

**Minimum Operational  
Performance Standards (MOPS) for  
Strapdown Attitude and Heading Reference  
Systems (AHRS)**

RTCA DO-334  
March 21, 2012

Prepared by: SC-219  
© 2012 RTCA, Inc.

Copies of this document may be obtained from

RTCA, Inc.

Telephone: 202-833-9339

Facsimile: 202-833-9434

Internet: [www.rtca.org](http://www.rtca.org)

Please visit the RTCA Online Store for document pricing and ordering information.

## FOREWORD

This report was prepared by Special Committee 219 (SC-219) and approved by the RTCA Program Management Committee (PMC) on March 21, 2012.

RTCA, Incorporated is a not-for-profit corporation formed to advance the art and science of aviation and aviation electronic systems for the benefit of the public. The organization functions as a Federal advisory committee, and develops consensus-based recommendations on contemporary aviation issues. RTCA's objectives include but are not limited to:

- coalescing aviation system user and provider technical requirements in a manner that helps government and industry meet their mutual objectives and responsibilities;
- analyzing and recommending solutions to the system technical issues that aviation faces as it continues to pursue increased safety, system capacity and efficiency;
- developing consensus on the application of pertinent technology to fulfill user and provider requirements, including development of minimum operational performance standards for electronic systems and equipment that support aviation; and
- assisting in developing the appropriate technical material upon which positions for the International Civil Aviation Organization and the International Telecommunication Union and other appropriate international organizations can be based.

The organization's recommendations are often used as the basis for government and private sector decisions as well as the foundation for many Federal Aviation Administration Technical Standard Orders and several advisory circulars.

Since RTCA is not an official agency of the United States Government, its recommendations may not be regarded as statements of official government policy unless so enunciated by the U.S. government organization or agency having statutory jurisdiction over any matters to which the recommendations relate.

This Page Intentionally Left Blank

## TABLE OF CONTENTS

1.0	Purpose and Scope .....	1
1.1	Introduction.....	1
1.2	System Overview .....	2
1.2.1	Attitude and Heading Reference Unit (AHRU).....	2
1.2.2	Magnetic Sensing Unit (MSU) .....	2
1.2.3	Option(s):.....	2
1.2.4	Complementary aiding devices:.....	2
1.3	Operational Applications(s).....	3
1.4	Required Functions .....	4
1.4.1	Attitude .....	4
1.5	Optional Functions.....	4
1.5.1	Heading.....	4
1.5.2	Turn and Slip .....	4
1.5.3	Other Functions.....	4
1.6	Operational Goals .....	4
1.7	AHRS Categories.....	5
1.7.1	Modes of Operation .....	5
1.7.2	Category Strings .....	5
1.8	Test Procedures.....	6
1.9	Definitions of Terms and Acronyms.....	7
2.0	Equipment Performance Requirements and Test Procedures .....	11
2.1	General Requirements.....	11
2.1.1	Airworthiness.....	11
2.1.2	Intended Function .....	11
2.1.3	Fire Protection .....	11
2.1.4	Operation of Controls .....	11
2.1.5	Accessibility of Controls .....	11
2.1.6	Effects of Test.....	11
2.1.7	Design Assurance .....	11
2.1.7.1	Complex Electronic Hardware Compliance .....	12
2.1.7.2	Software Compliance.....	12
2.1.8	Multiple Operating Modes and Performance Categories.....	12
2.1.9	Range of Operation.....	13
2.1.10	Magnetic Field Strength.....	13
2.1.11	Geographical Restrictions.....	13
2.2	Equipment Performance - Standard Conditions.....	13
2.2.1	General Requirements.....	13
2.2.1.1	Conditions.....	13
2.2.1.2	Stationary Startup .....	14
2.2.1.3	In-Flight or In-Motion Alignment .....	14
2.2.1.4	Annunciations .....	14
2.2.1.5	Output Characteristics .....	15
2.2.2	Attitude Performance Requirements.....	16
2.2.2.1	Equipment Categories.....	16
2.2.2.2	Static Accuracy .....	16
2.2.2.3	Dynamic Accuracy .....	16
2.2.3	Heading Performance Requirements .....	16
2.2.3.1	Equipment Categories.....	16
2.2.3.2	Static Accuracy .....	17
2.2.3.3	Dynamic Accuracy .....	18

	2.2.3.4	DG Mode Considerations .....	18
2.2.4		Degraded Mode Performance Requirements .....	18
	2.2.4.1	Incorporation of Degraded Mode .....	18
	2.2.4.2	Degraded Mode Accuracy (Pitch/Roll) .....	19
	2.2.4.3	Degraded Mode Accuracy (Heading) .....	19
2.2.5		Turn and Slip Performance Requirements .....	19
	2.2.5.1	Rate of Turn Performance Requirements .....	20
	2.2.5.2	Standard Turn Bank Angle Performance Requirement .....	20
	2.2.5.3	Slip Skid Performance Requirements .....	21
2.2.6		Aiding .....	21
	2.2.6.1	Aiding Source Performance .....	21
	2.2.6.2	Loss of Aiding .....	22
	2.2.6.3	Annunciated Faults from Aiding Source .....	22
	2.2.6.4	Faulted and Non-Normal Performance .....	22
2.3		Equipment Performance - Environmental Conditions .....	23
	2.3.1	Performance Standard for Environmental Testing .....	25
2.4		Equipment Test Procedures .....	26
	2.4.1	General Tests .....	26
	2.4.1.1	Conditions .....	26
	2.4.1.2	On Ground (Stationary) Starting .....	27
	2.4.1.3	In-Flight or In-Motion Alignment .....	27
	2.4.1.4	Annunciations .....	27
	2.4.1.5	Output Characteristics Tests .....	27
2.4.2		Attitude Performance Test Procedures .....	28
	2.4.2.1	Attitude Static Accuracy and Alignment .....	28
	2.4.2.2	Attitude Scale Factor Static Accuracy .....	29
	2.4.2.3	Multi-Axis Dynamic Accuracy Flight Evaluation .....	29
2.4.3		Heading Performance Test Procedures .....	30
	2.4.3.1	Heading Static Accuracy .....	30
	2.4.3.2	Heading Dynamic Accuracy, Category H1 through H5 .....	30
	2.4.3.3	Static Accuracy, DG Mode, Category H6 through H11 .....	31
	2.4.3.4	Dynamic Accuracy, DG Mode .....	32
	2.4.3.5	DG Mode Indication .....	32
	2.4.3.6	DG Mode Pilot Adjustment of Heading .....	32
2.4.4		Degraded Mode Flight Evaluation .....	32
2.4.5		Turn and Slip Test Procedures .....	32
	2.4.5.1	Turn Rate Test Procedures .....	33
	2.4.5.2	Standard Turn Bank Angle Test Procedures .....	33
	2.4.5.3	Slip Skid Test Procedures .....	33
2.4.6		Aiding Tests .....	33
	2.4.6.1	Loss of Aiding Source Data .....	33
	2.4.6.2	Annunciated Aiding Source Data Faults .....	33
	2.4.6.3	Un-annunciated Aiding Source Data Faults .....	33
	2.4.6.4	Test for Aiding Source Issues .....	34
2.4.7		Multi-Axis Dynamic Accuracy Laboratory Test .....	34
	2.4.7.1	Heading Modes, Category H1 through H5 .....	34
	2.4.7.2	DG Heading Modes, Category H6 through H11 .....	35
	2.4.7.3	Equipment with No Heading Output, Category HX .....	35
2.4.8		Flight Evaluation for Declared Performance Categories .....	35
2.5		Requirements for Display of Pitch, Roll, Heading and Turn and Slip Information .....	38
	2.5.1	General Display Requirements .....	38
	2.5.2	Annunciation .....	38
	2.5.3	Pitch and Roll .....	38
	2.5.4	Direction Indicator .....	39

2.5.5	Turn Rate, Standard Turn Bank Angle and Slip/Skid.....	39
2.5.5.1	Turn Rate .....	39
2.5.5.2	Standard Turn Bank Angle .....	40
2.5.5.3	Slip/Skid .....	40
2.6	Display Tests.....	40
2.6.1	General Display Tests.....	40
2.6.2	Display Annunciation Test. ....	40
2.6.3	Pitch and Roll Display Test .....	40
2.6.4	Direction Indicator Display Test.....	41
2.6.5	Turn and Slip Indicator Tests.....	41
2.6.5.1	Rate of Turn Display Test.....	41
2.6.5.2	Standard Turn Bank Angle Display Test .....	41
2.6.5.3	Slip/Skid Display Test .....	41
3.0	Installed Equipment Performance and Test Requirements .....	1
3.1	Equipment Installation .....	1
3.1.1	Accessibility .....	1
3.1.2	Aircraft Environment.....	1
3.1.3	Display Visibility .....	1
3.1.4	Dynamic Response .....	1
3.1.5	Interference Effects.....	1
3.1.6	Inadvertent Turnoff.....	2
3.1.7	Aircraft Power Source .....	2
3.2	Conditions of Test.....	2
3.2.1	Conformity.....	2
3.2.2	Power Input.....	2
3.3	Test Procedures for Installed Equipment Performance.....	2
3.3.1	Ground Tests.....	2
3.3.1.1	Equipment Function.....	2
3.3.1.2	Interference Effects.....	2
3.3.2	Flight Evaluation Procedures.....	3
3.3.2.1	Evaluation Procedure.....	3
3.3.2.2	Rotorcraft-Specific Flight Evaluation Procedures .....	3
Appendix A: Validation of Equipment Performance Using Simulation.....		A-1
A.1	Simulation Fidelity .....	A-1
A.2	Simulation Validation .....	A-1

## TABLE OF FIGURES

Figure 1-1	AHRS Diagram Example.....	3
------------	---------------------------	---

## TABLE OF TABLES

Table 1-1	AHRS Category String Examples .....	5
Table 2-1	Attitude Performance Requirements .....	16
Table 2-2	Heading Performance Requirements .....	17
Table 2-3	Turn and Slip Categories .....	20
Table 2-4	Turn Rate Accuracy.....	20
Table 2-5	Required DO-160G Testing By Category .....	24
Table 2-6	Attitude Static Accuracy Test Positions .....	28
Table 2-7	Attitude Static Accuracy Scale Factor Test Positions .....	29
Table 2-8	DG Mode Scorsby Table Test Parameters.....	35
Table 2-9	Sample Flight Evaluation Profile .....	37
Table 3-1	Rotorcraft Flight Maneuvers .....	4

This Page Intentionally Left Blank

## 1.0 PURPOSE AND SCOPE

### 1.1 Introduction

This document contains minimum operational performance standards for Strapdown Attitude and Heading Reference Systems (AHRS). This document is intended for equipment that does not use gimballed sensors. This document is intended for equipment that outputs attitude (pitch and roll). It also addresses functions of heading, turn, slip, and the display of this information. This document does not specify the format of data outputs. These standards specify equipment characteristics that should be useful to designers, manufacturers, installers and users of the equipment.

Compliance with these standards is recommended as one means of assuring that the equipment will perform its intended function(s) satisfactorily under all conditions normally encountered in routine aeronautical operation. Any regulatory application of this document is the sole responsibility of appropriate governmental agencies.

Section 1.0 of this document provides information needed to understand the rationale for equipment characteristics and requirements stated in the remaining sections. It describes typical equipment applications and operation goals, as envisioned by the members of Special Committee 219, and establishes the basis for the standards stated in Sections 2.0 through 3.0. Definitions and assumptions essential to proper understanding of this document are also provided in this section.

Section 2.0 contains the minimum performance standards for the equipment. These standards specify the required performance under standard and environmental conditions. Also included are recommended laboratory and manufacturer flight evaluation procedures necessary to demonstrate equipment compliance with the stated minimum requirements.

Section 3.0 describes the performance required of the installed equipment. Tests for the installed equipment are included when performance cannot be adequately determined through laboratory testing.

Appendix A discusses demonstrating equipment performance using simulation in lieu of flight evaluation.

This document considers an equipment configuration consisting of an Attitude and Heading Reference Unit (AHRU) with complementary peripheral equipment specified by the AHRS manufacturer.

Operation performance standards for functions or components that refer to equipment capabilities that exceed the stated minimum requirements are identified as optional features.

The word “equipment” as used in this document includes all components and units necessary for the system to properly perform its intended function(s). For example, the “equipment” may include an AHRU, an associated mounting tray, and a magnetic sensing unit. In the case of this example, all of the foregoing components and units comprise the “equipment.” It should not be inferred from this example that each AHRS equipment design will necessarily include all of the foregoing components or units. This will depend on the specific design chosen by the manufacturer.